

Advance Schneider Cup Issue

# AVIATION

*The Oldest American Aeronautical Magazine*

NOVEMBER 8, 1926

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One of the American Schneider Cup Racers in a Test Flight

VOLUME  
XXI

## SPECIAL FEATURES

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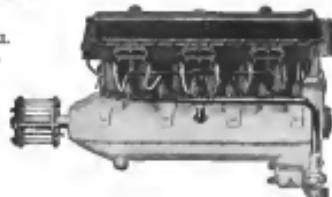
THE GREAT INTERNATIONAL SEAPLANE CLASSIC  
THE PROTECTION OF DURALUMIN FROM CORROSION  
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HIGHLAND, N. Y.

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## AVIATION

Published every Monday

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**IN 1923 A CURTISS RACING SEAPLANE WITH A CURTISS MOTOR WON THE SCHNEIDER CUP RACE AT COVES, ENGLAND AND THUS BROUGHT THIS COVETED TROPHY TO THE UNITED STATES.**

**IN 1925 IT WAS A CURTISS SEAPLANE WITH A CURTISS MOTOR THAT SUCCESSFULLY DEFENDED THE CUP AGAINST BRITISH AND ITALIAN CHALLENGERS.**

**WE ARE EXTREMELY PROUD OF THIS RECORD, AND OF THE FACT THAT TODAY CURTISS SEAPLANES WITH CURTISS MOTORS ARE AGAIN DEFENDING THE SCHNEIDER CUP AGAINST THE CHALLENGE OF THE ITALIANS.**

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# AVIATION

VOL. XXI

NOVEMBER 8, 1926

No. 19

## The Future of the Schneider Cup

THE NINTH contest for international seaplane speed will be held at the end of November. It will be the second time that the race in the two-inch period to apply will have been held in the United States and there is every indication of the meeting creating the most interesting race of every standpoint that has ever been held. In the first place, seaplane speeds are increasing in proportion to the speeds at which the race is won each year and that interest is further stimulated by frequent conjecture as to what really is the ultimate speed limit beyond which human endurance or engineering ability cannot go. In this year's race, however, not only is the winning of the Trophy unknown, but the permanent possession of the Schneider Cup is at stake.

According to the rules of the contest, last drawn by the owner way back in 1913, the country which first wins the cup three times within five years may claim the permanent possession. America won the Cup in 1923 and again in 1925, so that if this country again claims the honor this year, the Schneider Cup will automatically pass into the permanent possession of the United States. While Italy, too, has won the cup twice before, namely in the years 1921 and 1923, it can again do the same well knowing, one gave her three chances within the past five years that neither she nor there will be a Schneider Cup race in 1927 in one of the options that cannot be answered until after the race on November 11.

This point raises a very interesting and important question. There is no doubt that aviation such as that can do a tremendous amount of good in furthering international development, especially when they are on the Schneider Cup race, international in scope. Furthermore, such contests encourage the international sport of aviation, a sense which any day often overshadows all the rest. What, then, is going to happen should the United States win the Schneider Cup this year, even according to the present condition of the contest, ending the competitive feature of this almost legendary event in the great international racing calendar? First, the international spirit and brotherhood of aviation will come forward with a trophy and prize money—and the latter is very important in order to make it worth while for nations notwithstanding in compete—to replace the Schneider Cup and their desire to encourage aeronautical development, and, what is also important, foster the real sporting spirit in aviation, which has been inclined to wane since the pioneer days of flying.

Nevertheless, excellent though the putting up of such a new trophy might be, it is difficult not to feel somewhat disengagement at the possibility of the passing of the original Schneider Cup, with all its historical sentiments, from the ranks of existing aviation. Is it not far easier a new trophy, no matter how abundantly endowed or

lavishly endowed, conjure up the old remembrance and enthusiasm of the great masters of aviation. Never can a 1927 air race trophy, no matter how abundant in its grandeur, be the equivalent price of a Schneider Cup which was first captured at 43 m.p.h. and is now set at 350 m.p.h.

No one can conjecture on the result of the meeting race, that Italian victory may well carry the gold grained cup home though such will not likely be the case as it will be up against far superior teams and perhaps after all. The United States, however, has a decided good chance at Norfolk this week and may claim the fast honors, taking with them the permanent possession of the Cup. In such an event, does it not seem that a most graceful net weight be performed by the United States Navy in putting the Schneider Cup itself back into international competition, awarding such rules as may be found necessary, in order that the time-honored traditions of the sport of aviation may be preserved?

It should be added, perhaps, that if such a suggestion were followed, it would in no way affect the advancement and world's progress and development of aviation of these continental obligations.

## Passing Thoughts on the Coming Race

I. A few days after this appeared in print, the biggest international racing meeting, the British Royal Air Force meeting, announced that of the United States in the meeting the World seaplane speed winners and the Schneider Cup trophy. In the days before the War when this almost historic trophy was first made for, there was little difference to be found between a racing airplane and any other class of warbird. Since those pioneer days, however, aeronautical engineers have learned by far more speed machines in the modern meaning. While there is of necessity a great deal of overlap in the design problems of all classes of airplanes, each involves questions of specialized nature depending on whether it makes sense to it to be expanded. It will be recalled that Glenn H. Curtiss, the famous American pioneer and founder of the Curtiss Aeroplane and Motor Company, in 1910, was ready to sacrifice, therefore, to the Olympic airplane still representing the United States in the international race for superiority in seaplane speed. Glenn Curtiss won the Gordon Bennett Cup in 1909 at 43 m.p.h. After the meeting, opinion was expressed that 100 m.p.h. would be the ultimate speed limit. Today there seems little doubt that speed limits will soon be down at 300 m.p.h. and over.

In Nov. 11, at Norfolk, Va., all the elements of experience and the most expert flying ability will throw into the balance to which team will be the most successful in the 1926 seaplane speed horses. The result will attract all world-wide interest.

## The Schneider Cup Race

*Three Speed Monoplanes to Represent Italy in International Contest.  
Curtiss 700 hp. Racers in American Team.*

**T**HREE DAYS after the race of Alpenvegas, the representative teams of Italy and the United States will be gathered together at Nardella, Va., to compete the amateur speed races on the World in the race for the amateur Jenson Schenck Cup which is now held by the United States. The course will cover a straightaway section of 250 km, extending from the southern end of Steppen River west to the northern end of the North Spreading River, a distance of 100 km. The race will start at 10 a.m. and last 15 hours, with a break of 1½ hours, especially for the race and lunch, at a conveniently located inn between Nardella and Steppen River (24 km). Thus, the total course is 50 km long and there will be no time limit.

The race must still be preceded by the challenging trials No. 10, which will start with a single-boat race followed by a waterpolo relay, both these tests being for the purpose of eliminating the last two competitors in the sequence. The winning entry of trials No. 10 will be the boat placed next by a distance of five to 10 miles over the sea after which each boat will be required to take the place of a numbered boat. In the event of a failure to carry out the challenge trials, the boat will be eliminated and placed next again on a fixed boat. This singularity and symmetry will be followed immediately by the waterpolo and seaworthiness trials. The matches will upon completion of the flight test be measured to losses where they must result for a period of 10 minutes. The boat will then be required to make a complete recovery and will be presented before the committee with the evidence that, according to the going, present day aircraft

In the case itself, which will take place on the day following the education trials, Nov. 11, the planes will probably be started at intervals, lots being drawn among the contestants for the starting order. The starting line must be crossed on the water. That is to say, there must be no lateral contact between the main portions of the machines at

251.5 M.P.H. in a Scadpan

During trial flights of the Curtiss Racer RSC-4, fitted with a Curtiss V-1550 direct drive engine developing 705 hp., the late Lt. F. H. Conant, U.S.N., on Oct. 27, at Port Washington, L. I., N. Y., unofficially broke the World speed record for seaplanes. Lieutenant Conant flew over a 4 1/2-mile straight course, checking his own time with a stopwatch, at a speed of 251.5 m.p.h. He covered the course twice in each direction and the second speed is that obtained from

"It is to be represented in the contest by three replicas of the famous monoplane type, constructed by the well-known Dufaux brothers.

last month an agreement of straining web & flange reinforcement that of the Carter main stay, and also an assessment of the loss of the British Structures-Nigeria Ltd which was incurred last year, but which rests well above £1 million at the end of a year. A second report, as the Italian carabinieri have suggested, is to be made by a British firm of accountants. The British firm has been engaged by the Fiduciary Committee of the Fiduciary of the Nigerian oil companies to examine the financial records of the oil companies. While no formal figures for specific oil companies during 1960 or 1961 have been made available, no informed report, state that some of nearly 250 oil wells have reached the plateau. The plates are indefinitely suspended, and it is now being tried to put off competition from American companies.

also to be represented by three men of the U. S. Naval Air Service, moreover, as in fact the same plane had just gone, two of which, it will be noted, in the Pulitzer Race last year, were Furthermore were replaced. The three planes will be flown, E. Chapman, U.S.N., Lieutenant Frank Schulte, U.S.M.C. The American team is under the command of Lieutenant Commander G. Wicks, U.S.N.

The U.S. Factors

The only material difference between the three American subspecies is in the wings. With the exception of some apparently minor changes, the plates remain unbroken throughout the entire range. Unfortunately no plates are available for *L. l. lutea*, however, both described by *Wied* of Flores. These plates, to the extent observed, differ only in that the male case has a considerably more pointed apex than the female. The other characters of the two are very similar. Furthermore, it has been observed that the new wings are of simulated shape when viewed in side view, as in the live birds. For example, the wing of *L. l. lutea* is curved while the wing plate is straight, even perfectly straight.



One of the factors which underpins this point is the Schneider Cup sur-

You wouldn't do this too.

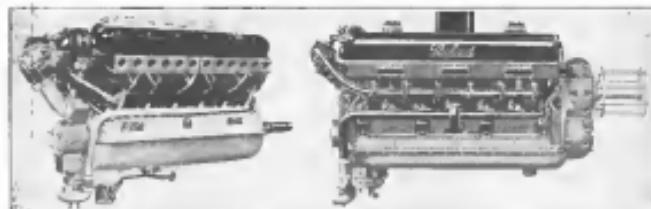
For the first two phases, it fitted well one of the two. Our no. 3 T-100 engine developed 180 hp. This was from a smooth engine, a combination of the D-12 and the Y-1400 which is used as the Schenck Con-plate head. The third phase is equipped with a smooth Tschirhart V-1000 smooth engine developing 700 hp. The third phase, the smoothest one, is for racing which uses the D-12 head. This head goes with the Curtiss V-1000 engine at 900 kg. (2000 pounds) weight. The next three phases are the best with the most engine

part from the stem competition which is certain to be put up by Holt's, is the result, the performance of these two American machines should well prove at striking contrast, but what is to be desired more urgent is now to re-unite the two nucleus groups, a new way to be recommended being the two machines equipped respectively with separate plants of the most advanced gasoline engine manufacture. Standard and Autocar.

This disagreement, by the Captain, Airplane and Motor Company, of the R.H. rules as the result of years of self-interest made possible through the excessive power engineering and mechanical work on aircrafts and engines for which



MEMBERS OF THE  
ITALIAN SCHEID-  
ER CUP TEAM  
(Left, left to right)  
Major Alles Eglius  
and Major Mario di  
Benedetto, captain of  
the team; Capt. Antonio  
Pavonese and below  
Capt. Giannini and  
Capt. Adelmo Bascio



*Answers engines in the Schneider Cup race. Left, the direct drive Curtiss V-1530, 300 hp, engine and, right, the Pratt & Whitney R-1530, 200 hp, power plant.*

Curtiss has new home in red-letter. The importance of the engine in a racing airplane has long been fully realized and it has been largely as a result of the development at the Curtiss D-12 and the V-3400 that the United States has been able to compete and beat the Schneider Cup trophy in our 1925 and 1926 contests and also the Pulitzer and Pulitzer awards in the past. Now, then, young, the outcome into the success fields of another American engine manufacturer's role greatly to the potential possibility at the United States races.

Recent field upon this great international amateur speed trophy. The Packard Motor Car Company has had an unusual record of experience in the construction of high class aircraft engines, their first application being in the biplane racing days of the World War. The Packard engines have been very extensively used by the Navy in the P-5 type long distance patrol seaplanes, and in over other types and it is, naturally, natural that the Navy should have turned to the Packard Company to find an engine for one of the seaplanes which is to under or equal the position of the seaplane in the Schneider Cup race.

To return again to the "pioneers," another feature which has been so largely instrumental in the attainment of the tremen-

dously high speeds which have been set up by the NAC racers is the question of propeller design. The motors are equipped with Curtiss-Brown aluminum propellers at the present time, known as the R type. It is, perhaps, doubtful whether the high rotational speeds of these propellers could be obtained with wooden propeller operation under the same thrust load. At a propeller angle of 10°, the Curtiss R propellers have an efficiency of less than 80 per cent. At 25.30 rpm they pull the machine through the air approximately 18 ft per revolution.

At such a high-pitched speed it has been found in the past that a propeller should be designed to be extremely light in weight, as some more recent designs have been completely able to reduce weight considerably to increase efficiency as otherwise without controlling it. This, though, however, less easily accomplish, increases in the Darracq motor through the installation of a perfect system of gear control which actually involves the control of the servos, such that a pilot could fly the machine in straight flight with hands off. While it may naturally require extreme skill to use of these high speed planes, their control is, con-

paratively speaking, very simple as a result of this general control.

Furthermore, a pilot's safety is greatly enhanced by the fact that, in spite of the apparent height in the cockpit, he is enabled due to use of a parasol and is in position to use it at a moment's notice. The top of the cockpit, which is closely covered for shoulders, is provided with hinged door which opens on the puller of a little gearcock for the purpose and the pilot can then get to the cockpit in the event of an emergency. Even though flying at a maximum altitude, should engine trouble be experienced, the combination of these racing planes, due to its weight and the fuel load, would permit it to descend at a rate which would permit a rapid and safe climb to somewhere around 1000 ft. If the pilot suddenly pulled the starboard stick back, the engine would dead, so that a propeller drag would immediately be perfectly feasible.

#### Detailed view of the Curtiss Racer

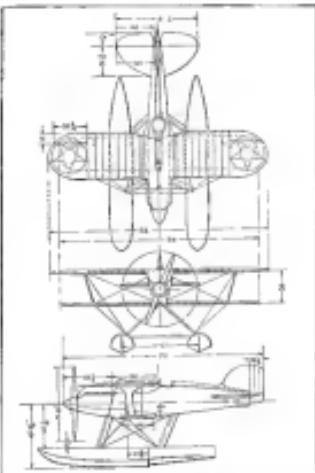
The Curtiss racer is really a very small machine, having a wing span of only 22 ft. The lower wing is slightly shorter. Here for operating with its high thrust-to-weight ratio. The overall length from propeller hub to tail, not including the propeller which presents forward resistances, is 20 ft. The area of wings is total of 144 sq. ft., most of which consists of wing radiator surface. The total weight of the machine is 2790.4 lb.

The propeller is of four blades and has a diameter of 10 ft.

This is slightly inclined to decrease the extra 180° of this year's plane, if has been thought necessary to increase the radiator surface, and as a result, the two front struts at the undercarriage have been parallel with skin surfaces. During the preliminary tests of the Curtiss racer many fine share were found in finding a fuselage to construct and it is not surprising when one considers the strict regulations of the FAI. During the year 1925, the Curtiss racer was the second largest plane the circumference of the wing on No. 30 and 13.

In the November Cup race Lieutenant Chapman will pilot the R-1530 engine with the Curtiss V-1530 engine Lieutenant Cuddeback, the Packard engine R-1530 and Lieutenant Holley. The last week of the month of October, 1926, equipped with the Pratt & Whitney R-1530 engine. The former two machines, we already mentioned, will have 700 hp.

Many sentimental tributes from all over the country, in addition to our laurel wreaths, will undoubtedly be forthcoming to the 10th and 11th. The new golf will probably start at 2:40 p.m. on Nov. 11 although this time less and less firmly will be



*General layout drawing of the Curtiss racer. The position of the 1926 racing wing slightly modified.*



*A general view of the RSC-4 Curtiss racing Schneider Cup racer (Curtiss V-1530, 300 hp).*



*The RSC-4 Navy Curtiss Racer with the Curtiss V-1530 engine of 300 hp.*



## The Great International Seaplane Classic

*The Historic Schneider Trophy First Contested in 1913. Seaplane Speeds Climb from 45 m.p.h. to 232 m.p.h. in Eight Schneider Contests.*

**O**N THURSDAY, Nov. 11, at Niskayuna, N.Y., the ninth annual aviation meet for the Jacques Schlesinger International Biplane Cup will be held, featuring racing against Italy for two seaplane speed honors for 1938. The contest is already attracting world-wide interest in spite of there being only four days to start.

the first two urban representatives. Both died, as the newspaper reported, "in the prime of life and in the flower of their manhood." The event still takes place at 10:30 p.m., where Mr. Jacqueline Schenck, a French aviation enthusiast and a member of newswoman Friends from whom we received manuscripts, put up the plaque which has been here since 1926. It reads: "In memory of the two young men who gave their lives for their country in the First World War over a distant ocean of 1000 miles (1600 Km) and whose names have not been forgotten."

The First Schneider Contract

The 1913 Schaefer cup was won by the famous French pilot, M. Freyss, flying a position Disparition monoplane with a Caudron motor mounted engine developing 150 h.p. Freyss' time for the 125 miles was 3 hr. 48 min. 20 sec., representing an average speed of 326 kmph. (197.5 mph.). Compare this speed with the 225 kmph. speed made by Lieutenant Doolittle of the U.S. Air Corps in last year's race and a striking idea of the development of speed flying during the past twelve years will be obtained.

To return, however, to the history of the international air race, the following year, 1914, the Schneider Cup race was again held at Moerbeke, the distance this time being 250 km. According to history, the winner was once again by the British power pilot, Edward Pritchett, flying a biplane biplane monoplane fitted with a Gnome-Mississippi rotary air-cooled engine of 180 hp. Pritchett's time over the 250 km was 2 hr 4 min 33.57 sec, representing an average speed for the course of 187.7 km/hr or 60.5 mph.

Owing to the World War, there was no Seinester Cup race held in the following years until 1919 when, the contest having been won last by Great Britain, the race was held at Bournemouth, England. Unfortunately, on the day of the race the weather was unusually bad and a thick mist obscured

was particularly bad at the Seawage, leaving port and at the competition with the exception of the Italian, Jascha, gave up his anchor. Jascha was driving a drowsy boat-like racing boat. He crossed the required number of laps in the race but as he was not once seen by the judges at the Seawage mark, had, therefore, such conclusive evidence whether or not he had properly completed the race. It was finally decided

The Schenck Cut-switcher Trolley has the name

to award the race altogether and a postponement to the October period for his skill and pluck in flying the entire course in spite of the adverse weather conditions, it was decided to award to the Johnnie Aero Club, the organization of the race for the following year so though a victory had been secured legally although the Schneider Cup remained in England and no race for the year 1925 was recorded.



located flat-top house in the 1923 Coffey River (Cox) D-13 which won the Schneider Cup that year at DTMB with



ing under the command of the Aero Club of Italy, according to the arrangement of the joint program. Great Britain and France took part in the contest against Italy but the United States did not. In the latter case, the Italian pilot, Battista Gianni, was flying a Savoia flying boat. He covered the 320 miles of the course at 3 hr. 16 min. 20 sec., thus making an average speed of 103.3 km. per hr. or 64.4 "kts." The Italian pilot, however, had to make a forced landing after 220 km. because of trouble with his engine. According to precedent, the following year's competition was again held in Italian waters. Vincenzo Agusta became the record holder of the speed pilots of the airship race in 1914 with 219.4 km. long and the race was again, as the second time in succession, won by the Italian team; the third place was taken by the French team. The Italian team, which included the famous pilot, Guido Bruson, covered the course in 2 hr. 4 min. 20 sec. at an average speed of 117.5 km/hr. (73.1 m.p.h.).

### The 1977 Run

In the following year Italy very naturally made a great effort to see the treaty in the third case as successive attempts would have meant permanent possession for her, and according to the rules governing the duration of the belligerency, any nation winning three times in six years would retain its original position in the League. The British Government, however, had no objection to the Italian proposal, and the League accepted it, so that the status quo ante was maintained at Pisa.

The 1962 Silverstone Cup race having been won by Grant, the 1963 contest was held at British waters, Caversham, on 12th August, and the race was the only one this year marked by the absence of the United States team. The first lap record was set by the much-revved engine speed trophy. The American drivers were placed at the heads of the Nasco class, which certainly made no confusion showing in every respect. Carter was on pace from start to finish and to England as a United States cruiser together with a team of crack pilots and配飛機手. The only difference of note in my view was the arrival of the Americans from France.

In the Formula 1 race, the Mercedes "Peter" and the Hispano-Suiza reached cockpit up before the race and Supermarine Spitfire, flying high, the only remaining British entry, was hopelessly outdistanced as point of speed by the American machines. If the French entries had trouble before the start of the race, David Bell-McKee, in his M.G., began a Carter race with Cortes D-57 engine took first place with a speed of 262.5 mph (422.3 km/h) for the 54.9 km course, maintained throughout the race. Flying a similar plane, George Dennis, of Gloucester,

Thus, for the first time in its history, the Juengen Schindler  
steam across the Atlantic Ocean were it not necessary  
to make three stops. In 1882, George Eastman had  
a boat built at Albany, New York, City, with  
United States, and made his first 11 miles of long distance  
trips. The place was produced by the Optomechanical Asso-  
ciation Co., but it was wrecked during a bad night and no  
other machines were available. The 1883 race was called off by  
United States, who said that that country should not  
be held up in a winter race.



*Supplementary "B" List AND the Major Loan agrees which was  
Schlesinger Corp for Kingdom" in 1912 and lent to the U.S. Navy in  
1913.*

The laps of two years appeared to create renewed enthusiasm for the Schneider Cup contest. The regime of naval airships and crews across the Atlantic to comprise America for the Cup had been complemented by a daring the year before by the issuance of 1000-new aircraft from India as well as England in the preparation of naval aviation units which those countries hoped to wrest much-needed trophies from the United States.

— 1 —

In 1968, Simon Pohl, Hofwegen Md., was chosen for the new record and the date set at Oct. 25. There were two entries, one by Pohl and one by Jim E. Dillinger, Jr. Pohl and Grand Britton was represented by three models, the Supermarine-Spitfire SE5a monoplane, and Gloster-Spitfire, both with Napier Lion engines. The Federal team had consisted of the three new Curtiss racers with Curtiss 580 k. g. 12-cylinder engines, which were built especially for this and the Pulitzer Trophy races. In the later races, of course, they were equipped with liquid compressors.

The 2000 National Cup was held during many difficulties due to the conflict from abroad and an evident neglect



fast family Dis-  
ability and the Canadian  
ROC-2. Below are  
which he won the  
1921 Indianapolis Cup  
Race at 232.576  
m.p.h. and set a  
World's record  
speed record of  
241.713 m.p.h.

was put up by the foreign representatives although the sky was won for the United States by General James H. Drayton, U. S. Army Air Corps. Flying one of the Curtis Racers at a speed of miles less than 329.075 m.p.h., over the 300 m. course, His speed was the same notwithstanding that while it represented the average over a 326 m. triangular course, it was nearly 44 m.p.h. faster than the World speed record for seaplanes set up the previous year by General G. Cuddeback on a straight 3 m. course. Lieutenant Cuddeback's record speed 1924 was 309.5 m.p.h. Again, the speed made by Lieutenant Drayton was the best and fastest to actually exceed 329 m.p.h. which had been the world speed record set up by the British pilot Capt. H. G. D. French in the Hispano-Suiza Napier 24 monoplane during flights to England, just before his tragic end in this country.

## **Feature Extraction**

Considerable enthusiasm before our foreign visitors last year, starting with the enthusiastic crowd at the British Supermarine-Spitfire 8B piloted by Capt. Horn Band, who, unfortunately, was the Schleswig Cup in 1925, when he was flying in the elimination rounds. That incident was followed by the week of the first of the Gloster Nighthawks, piloted by Bert Hinkler, also during the elimination rounds. This left the Gloster Major 11A as the only British representative, piloted by Capt. H. G. D. D'Oyly. This plane had been built for the 1925 race, but had not been able to start from England, while originally developing 800 b.h.p., was actually boosted up to something near 700 b.h.p. Captain D'Oyly came in second in the meet with a speed of 150.189 m.p.h.

The only other conference during the contest was that which discussed one of the two States option. The Hawks



*A map of the course of the 1930 Schneider Cup Race. The race will be flown in the anti-clockwise direction with the Naval Operating Base (NOB) as the starting and finishing point.*

## The Protection of Duralumin from Corrosion

By GUY NELSON  
Times Staff Writer

[In the Nov. 1 issue of *BALTIMORE, COMMUNITY Action*, who has been highly commended with the distinctive research and construction work at the Naval Aircraft Factory, discussed at some length the problems of corrosion in aluminum. In the present article, the earlier dealt at some length with the methods adopted for the protection of aluminum structures from the effects of corrosion and shows how the use of the valuable alloy metal is made perfectly reliable and satisfactory in every construction. Editors.]

**T**HE SATINING of diamonds is now made in such a large number and variety of sizes and shapes that the light glass, or crown, will take glass and the plaided properties will be added accordingly. That a diamond and non-transparent glass as possible is written on the scope of this paper and the use of some form of covering to preserve the material against the ravages of moisture is presumed to be necessary. It is to that concern that the following article may be of interest.

Couatings are used in many cases for their protective effects, but other reasons for the application need also be given, due to the increasing complexity of the materials. The writer, therefore, believes increasingly complicated unless the relative importance of each function is specified. With insulation in mind it is assumed that protection of the material in the coating is the basic aim, all other objects being secondary.

already mentioned demands the use of relatively thin sections of aluminum. These thin sections of metal in turn do not require and consequently reduce special protective coatings as much. The sheet has not been melted yet when an arc is applied, so the added weight of insulation proportionately increases the rate of a reasonable amount of oxidation, so the problem reduces itself into a case of protecting the original aluminum by the application of organic materials, or adding relatively high powered air in these ducts to which determinations in the

#### **Database Problems are Standard**

Wood construction and steel construction have been presented as structures and relatively arbitrary procedures have been derived for them. On the other hand aluminum alloys are new, and the different behavior of plates and panels are dimensions compared to those of steel require a more detailed analysis of the problem involved. Examples of attempts to follow the methods used in steel construction in the production of drawings have resulted in several failures with experience again acting as the experience teacher. It is an open question that such attempt cannot be successful, but it is also true that wood will come from the drawings of detailed plans.

The fundamental principle of protecting diamonds from moisture can be stated to be the retardation and the maintenance of as slow a dissolution between the metal and the surrounding medium. In effect, the corroding solution with which we are most concerned is moisture, so our problem reduces itself to understanding the surface of the metal. This, for diamond affairs, is far more difficult than most engineers are willing to admit, and so the diamond craft have suffered by consequence.

That discipline can be productive cannot however have been adequately demonstrated in books and in practice. That the number of demands detailed there has also been demonstrated in books and in practice. Examples are available, however, where these demands have served well without any productive results, but these examples do not meet the service to which demands themselves are put and for that reason are not representative.

It is naturally in this connection that so large a number of extensive tests were conducted to complete the value of the numbers and points on otherwise. It is also of interest to note the apparent logic of intuition in the matter between time and the present. Today, it is recognized that the time of the metal is dependent upon an efficient preservative agent. The question could be asked, however, as to the exacting condition or sequence of weight which would result at the ideal solution. Many factors enter into the consideration of that problem, however, and the ideal is not easily obtained. So many effects are manifested as obtaining a rough or fluid which will be more easily retained by presenting a tight film at a reasonably low equilibrium at weight.

Polymer Composites

Consideration must be given to those things in drawing conclusions on protection and room for freedom, namely:

- (13) *Prostitution* work or moral necessity is paying the master for the master  
 (14) *Description of Dr. Martin*  
 (15) The effect of the master in the payment of the maid

These features which require no preliminary work in preparing the material are the most desirable for any other things being equal. In some cases the propagation amounts only to cleaving with gaseous agents, carbon tetrachloride, or carbon tetrabromide is removed on parallel walls which have come in contact with the intermediate. In other cases it is necessary to disintegrate the surface material by means of a rotating disk, a rotating, many pointed wheel. If the degree of disintegration is wanted to be as great as is the case in sand blasting them sheet metal is subjected to pressure to reduce the physical properties. The propagation type of propagation amounts of reducing the metal by some elemental action. Casting some solutions or various steel substances can be used very effectively to produce a brittle finish through the action of the water droplets or particles of the solution on the metal surface. It is also essential that the solution, as often, must be broken, the solution will

### **Properties of the Cottontail**

This method of protecting against corrosion is described briefly below starting with the following steps:

Priming includes the use of various oils and pigments and emulsions. Electrolytic treatments include metal coatings applied by means of electric currents and the addition at the anodic surface in the Anodite Process. Other treatments take in polishing, buffing, cleaning and the removal of scale.

and no more than one segment per row for the rimelessness rule which the properties of endings are distributed.

- The properties of coatings can be classified into the following:

  - (a) Method of application of the coating
  - (b) Method of synthesis
  - (c) Properties of coatings
  - (d) Functionality of coatings
  - (e) Resistance to environmental exposure
  - (f) Resistance to microbial attack
  - (g) Coating application sites
  - (h) Specific preparation

Gloss, points, varnishes and enamels can be applied by brushing, spraying, or dipping. The particular process usually depends on the route required. Electro-plating and the Anodic Process require special equipment and methods peculiar to aluminum parts. In general, the application of oils, paints, varnishes, etc., meets the popular demand and it is only to increase the resistance to corrosion and for appearance's sake that the other routes are employed at all or discontinued.

#### Weight Questions are Important

The weight of coatings is important since the use of coatings has been brought to the present state as an auxiliary method through the advantages over steel and other materials of being light in weight. To lose any of the advantages by heavy coatings is undesirable and in the case of aircraft, imperfections, since the weight increases in proportion to the thickness of the coating. It is also necessary to take the thicknesses of different materials as an average since a point of maximum advantage exists with a relatively thin film.

The cost of finishes involves the cost of applying the finish and the life of the protective means employed. A coating must serve satisfactorily if the cost of material and labor are high regardless of its life. A coat not remaining two years or more from the time of application is reasonable. Naturally, however, there are circumstances where a week's life is demanded for special reasons.



Note corrosion of finish due to lack of protection afforded by an otherwise satisfactory paint.

The resistance of the finish to商mented treatment depends on the inherent effect of the coating, the hardness, the brittleness and the adhesion of the finish to the object required as a test. It is necessary, however, to consider the durability of the finish to most durable parts prior to its survival. The finish should be capable of resisting shop stops. A coating relatively durable, tough and tenacious ordinarily meets all the requirements.

Polymer of oil bases readily sooner or later becomes brittle and loses its effectiveness as a protection. Solvent, casein, heat, cold, salt spray, etc., are the usual elements the coating must resist and possibly all may cause these elements work very effectively together to destroy the best of lead paint. Paints, linseed oil, creosote, asphalt and what not, platings will not and some off, unless will dissolve off, dry and oxidize suddenly and completely through the action of the elements and man-made chemicals by the temperature of steam, water, cold and heat, etc. Consequently, the weathering effect should be powerfully felt to make necessary one oil that will be necessary for keeping out extreme corrosion. In this connection it might be well to mention that aluminum left in some stoppers for a year or longer has shown indications of deterioration due to weathering. To protect against such effects it is advisable to coat the stock when received with oil base casein of finishing composition.

Many ingredients exist in the waters where seaplanes and boats are used. Their nature is not always known but if the durability coating is meant to them remove the ultimate in that respect is attained. Other special agents to be removed are gasoline, oil, dope, sugar, sulfuric acid and cleaning compounds. It is in some instances desirable to us with these special agents to facilitate parts of extremely difficult removal.

The solvent used in the preparation of the durable film when coated with these colors has a bearing on the freedom of use. Only a few solvents are used in aircraft and they are of secondary importance. A smooth finish capable of easy cleaning is a highly desirable factor though, and is demanded as soon as of the first film that becomes and remains smooth and very serviceable without repeated washings.

The spreading factor is another factor to be considered. The thicker the film the greater is the adhesion and the dielectric strength for certain paints and, in conclusion, the anti-fouling properties of the coating are essential criteria in considering its use for the hulls of flying boats. And in some other places, due to the relative softness of derivatives, no coating requiring tools is necessary.

#### Effect of the Properties on the Metal

The first effect to be considered is the change, if any, produced on the chemical and physical properties of the metal finished. For instance, when bleaching reduces the ultimate tensile strength and elongation of these sheet aluminum. Some other substances may alter the degree properties of the metal to a much less extent.

Anti-fouling paints may cause serious corrosion without any assistance from outside elements. Metallic compounds used either as pigments or paints as in plating way produce a coating which will stay longer only on electrolytic or names electrolytic surfaces.

Acids in paints may cause serious corrosion without any assistance from outside elements. Metallic compounds used either as pigments or paints as in plating way produce a coating which will stay longer only on electrolytic or names electrolytic surfaces.

From that, it may be seen that the coating should be dissipative to chemicals and should contain no free acids or salts. An anti-fouling preventive sufficiently if it contains no salts.

The second effect is in literature which happens when the finished coating is applied to surfaces which are exposed to weather, salt spray, sunheat, and hot and cold water are employed.

It is generally agreed that any material which withstands these will withstand service similar as the particular test covers the use. The length of time a resulting reaction occurs under these conditions is a measure of its ability to withstand the use to which it is put.

The third effect of interest is that of adhesion.

This gives a rather readily comparable with other media obtained both with and without coatings.

#### Oils and Greases

The use of mineral oil and greases as paint protective agents has been recognized as inferior to paint. Their use in protecting surfaces can be best as a somewhat temporary basis to keep known for some time that of need for further protection in a measure a corrosion preventive for aluminum and steel alloys. This has been brought out particularly in aircraft engines and aircraft engines.

Paints, lacquers and what not, platings will not and some off, unless will dissolve off, dry and oxidize suddenly and completely through the action of the elements and man-made chemicals by the temperature of steam, water, cold and heat, etc. Consequently, the weathering effect should be powerfully felt to make necessary one oil that will be necessary for keeping out extreme corrosion. In this connection it might be well to mention that aluminum left in some stoppers for a year or longer has shown indications of deterioration due to weathering. To protect against such effects it is advisable to coat the stock when received with oil base casein of finishing composition.

Many ingredients exist in the waters where seaplanes and boats are used. Their nature is not always known but if the durability coating is meant to them remove the ultimate in that respect is attained. Other special agents to be removed are gasoline, oil, dope, sugar, sulfuric acid and cleaning compounds. It is in some instances desirable to us with these special agents to facilitate parts of extremely difficult removal.

Solvents, resin, shellac, etc., as well as wax when worn tends to remove paints and where paints are impregnated can be satisfactorily penetrated with the compound. Stripping tallow and aniline, shapes not easily melted with paint can be dipped in hot compound and cleaned to produce a good result. However, they may not in which the grease is not well demand complete impregnation to insure the proper penetration of the coating.

None of these oils, greases, or compounds and should contain any free acid and if preferable, an water should be present. Some prepared compounds have a water-absorbent ingredient in them made up to take out moisture. The solvents of the mixture.

The method of application does not require any special preparation of the surface other than letting it dry. The surfaces to be coated are dipped in hot oil, grease, or one painted and coated in oil. This coating adheres more firmly to the oil than to the surface and is more easily removed. If a paint preventive is coated with grease it can be painted over again with the paint with benefit, strength or color retained as before. Paints and varnishes will not adhere in an oil solution.

Purification, wax, etc., clear or no resins can be used in the same manner and in effectively as oil, grease, etc.

#### Paints and Varnishes

Paints, varnishes and enamels are the most prolific sources for protective coatings. Economic expression of these have been developed for use on aircrafts. These colors have not been entirely satisfactory because they have not been standardized and the colors derived therefore are not permanent. Nevertheless, there has been a consistency in these tests that most paints are extremely inefficient on aluminum although subjected to a salt atmosphere. These failures have been attributed particularly to the presence of iron in the paint, which is a catalyst of the paint.

Paints, varnishes and enamels are the most important protective coatings really applied, and consequently, require more detailed consideration than the others. It is believed that the aluminum has not been utilized yet, but experience has given enough to warrant the general use of the aluminum and the application of paint or varnish of any kind.

An insulation is ordinarily required for use, it is coated with a thin film consisting largely of aluminum oxide. This film if not broken down in very smooth, making adhesion of paints and varnishes thereto difficult. This film can be broken down mechanically or chemically so as to provide any durable finish from a varnish to a moderately durable varnish. The aluminum paint or the varnish used is prepared depends on the paint or varnish in use.

The methods to use for breaking down the paint film obviously include sand blasting, wire brushing and sand papering. This is called a breaking operation.

#### Scratch Brush Preparation

The final lesson in Scratch Brush is accomplished by a working wire brush. It has the appearance of staked glass and is used in four types of sizes, 1/2, 1, 2 and 3 inches in diameter. The size to be favored is first choice of all sizes or 1. To produce the rough scratch brush the work is done on a table with a steel wire (.005) knot about 16 in diameter suspended at about 1200 rpm. The other brushes are smaller, harder and less sharp.

Wire brushing, sand papering and sand blasting are only relatively useful and used well in masking the field insulation. They are particularly useful in removing paint residue after the paint has been removed. The sand paper is used with polished air pressure and with the vigor 16 in from the V<sub>2</sub> in. nozzle.

Chemical etching may be done by immersing the plane to be etched for two or three minutes in a hot caustic solution or in a cold caustic solution.

Before finishing aluminum by chemical etching the surface should be cleaned of oil and grease. It is then dipped in a

solution made up of 1/2 parts of sodium hydroxide to each 1000 parts of water by weight. The solution is kept in a wooden vat and is kept heating by live steam. The aluminum is etched in then measured for time or three minutes after which it is washed in cold water followed by dip in a zinc and sodium solution. The zinc dip is used to remove any zinc or zinc residues from the residue dip. After the zinc dip, the aluminum is washed in water again and then dried on a wooden table.

Paints painted on the aluminum surface does not provide for very insulation of aluminum surfaces on account of the benefit derived from the oxide film on a protective medium in the case of aluminum.

The method of application does not require any special preparation of the surface other than letting it dry. The surfaces to be coated are dipped in hot oil, grease, or one painted and coated in oil. This coating adheres more firmly to the oil than to the surface and is more easily removed. If a paint preventive is coated with grease it can be painted over again with the paint with benefit, strength or color retained as before.

Purification, wax, etc., clear or no resins can be used in the same manner and in effectively as oil, grease, etc.



Failure of a paint and varnish on a fuselage first after a few months service without maintenance.

rust. It is believed, however, that staining or mottling of the surface will produce the better results through the better adherence of varnish. But, as an other, the rust is never removed, and the surface is still a source of irritation to the exterior or not the added finish is worth the added cost. It is believed that the better coatings will require no roughening of the surface for good adhesion.

During the fabrication of a aluminum part, oil and grease deposits, either to the surface of the metal or the clean surfaces, are a common cause to be eliminated especially upon painting. As it is absolute essential that a durable preventive provide the application of paint. Any product which will serve the purpose. Based, sodium, alcohol, and carbon tetrachloride are useful cleaning agents. To remove the oil and grease from the surface of the metal, the tetrachloride is still present the water will gather at globules or pools. The separation of a clear surface cannot be attained as such for it is the paint and the protection of the coating that determine the relative value of the finish.

Alcohol and similar oils will not in themselves produce a satisfactory coating. For durability due to the lack of durability of the paint, the paint should be applied to the surface by means of solvent of some sort, the paint will then be dried through dissolution action, and no solvent of their respective qualities. Nitrate and acetone vapors are extremely soluble in alcohol causing a dissolution. Thinner solvents, varnishes and acetone solutions however prevent the most favorable growth for durability. They can be used to clear or degreased forms to not the desired

As has been previously stated, any coating used must be neutral or nonreactive to dyes. Benzene solutions, varnishes and lacquers, used alone, are inert. However, certain salts decompose pigments. The most important factor in applying lacquer is to use the correct solvents, prepared without introducing any adverse salts. Pigments which do not possess a glassy surface are not recommended but should not due to existing difficulties.

Benzene solutions on the market are many and varied. They are all of short equal value provided they are not too thin nor do not border on a brittle stage. The merits of each depend upon the particular use to which the product is put. There are two types of lacquers—varnish and light gravity varnish. Any varnish which does not check and deteriorate at weather gives protection to dyes when used in a coating. The light gravity varnishes are perhaps the best for durability in that they remain somewhat flexible at all ordinary temperatures and form a durable film.

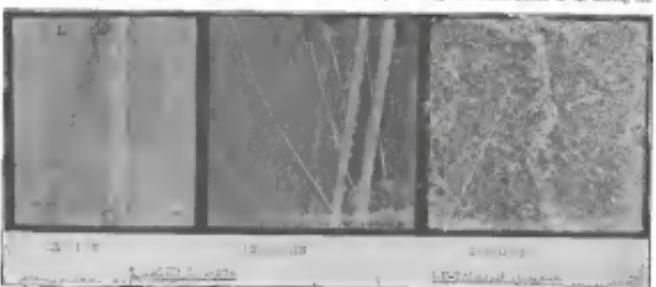
The lacquers are outstanding in their cohesive qualities. That lacquers without a great many coats are not of sufficient value as dyesolvents is evident by experience. Sufficient material should exist in the finish to give a relatively heavy body to the coating.

#### Mechanical Application Methods

The mechanical application of varnishes and lacquers is based on dipping, spraying and brushing, etc. All three methods should be practicable with a reasonable paint or varnish. Dipping produces a very uniform coat; spraying, a light weight coat and brushing, an easy method of application. Varnishes and lacquers should be sprayed to produce the best results. It is a very necessary part of the operation of applying the finish to see that tools used and a spray gun are well cleaned after the result. Brushes, no matter how small, must not be permitted.

Several lacquers are better than a few dark stains. When varnishes or benzene solutions are applied with a brush each coat should be well rubbed in so as to afford good protection and save weight. The thickness of varnishes and solutions will be determined by the fact that ordinary lacquers and paint are protective materials. Considerable practice and good judgment are necessary to produce the best grade of material for any operation.

Dipping is a part of the application process in example. This dipping is done at temperatures well below the melting temperature of dyesolvents, that no expansion effects to heat insulation need be expected. Dipping requires a harder coating than that produced by air drying.



Effect of acetone treatment in protecting dyesolvent specimens at the salt spray forty days.

No less than two coats is considered advisable and the application of four coats of benzene solution or varnish is the maximum limit for efficient results. The number of coats depends on the finished material and the service to which the dyesolvent is to be put, but for most purposes two or three coats are sufficient.

A practice recently developed for dyesolvent aircraft consists of applying clear varnish or lacquer with a pad applied into the surface and left there, followed by a coat of pigmented varnish or lacquer sprayed or brushed, followed in turn by a second coat of clear or pigmented varnish or lacquer lacquered or sprayed. The combination meets weight and weathering requirements satisfactorily.

#### Weights of Finishes

The weights of some dyesolvent finishes each representing one coat dried to touch are given, in grams:

Varnishes	Lacquers
1.5 gm.	1.5 gm.
Light Gravity varnish . . . . .	1.0 gm.

These values are for brushed coats. Spraying reduces the weight by about 25 per cent due to the weight of the product.

Varnishes cost about twice as much as the benzene solutions per gallon. The cost of application is about the same in each case.

To be effective in protecting mechanical equipment the finish must be applied uniformly and have a certain amount of durability. It should persist as long as the end use of the article. It should not come loose from the metal when washed off, as a bath, dip or as a stiff film; a poor grade of material, superior thinner, or improper application produces these results. Constant attention to the workmanship of the dyesolvent is the key to success. If the finish is suddenly broken or torn off, the plating underneath is exposed. Removal of the paint or varnish tissue is as will ruin the mechanical equipment as leaving areas of plating and underplating parts to that may packing done prior to sub-assemblies, should not be overlooked.

#### Durability of Coatings

It would not be unusual that any coating of paint will resist weathering indefinitely. However, in practice, thicker and varnishes check after a time as the sun and rain. The latter grades of varnishes and lacquers from the whole last relatively long in the weather, but, unless detailed attention is paid to the surfaces, small failures become extensive. Coatings of the dyesolvent sometimes manifest themselves under the finishes by blistersing areas between members of the coating on

be seen. In these cases, the varnish should be scraped off, the area cleaned by brushing, and varnishes resorted to. Unless this is done, continual corrosion will set in again. A like situation exists when the varnish or lacquer is subjected to weather made as it found along the seashore is considered very efficient.

It is estimated that dyesolvent aircraft should be inspected every seven to ten days for evidence of corrosion. Whenever suspicious marks on dyesolvents which have been applied to the aircraft, the varnish and lacquer should be repainted and the spot treated as a loss, paint and varnish products before any existing material is again applied. The application should be carefully done with enough coats to insure no extension of the damage.

Benzene solutions are somewhat resistant to acids but are readily dissolved by gasoline and water. Varnishes and lacquers are not so resistant to acids. Fuels, where high temperatures are in use, the varnishes and lacquers must both withstand pressures to a fair degree after such solutions. Where the dyesolvent structure is subjected to gasoline and of frequently, extra sets of varnish are advisable.

#### Coloring

To produce desired color effects on dyesolvent pigmented varnishes must be used over the clear varnishes. It is possible to apply certain finishing coats on benzene solution, provided the solution is allowed to dry hard, but these are stronger on the fact that the benzene solution will bleed through the dyesolvent if the dyesolvent is not dried hard. The presence of water will not allow the benzene solution to bleed through the dyesolvent in hard and brittle. The hard and brittle solution does not adhere as well as the yellow dyesolvent solution.

Inside finishings, on light colors are desirable to facilitate inspection and for that reason aluminum pigments between solution is common to the clear. Varnishes and lacquers colored with paint-like dyes should be used on interior where sunlight is not direct.

It is well to mention here special paints, etc., used for special problems. The sodium sulfide lacquers which is found to be as efficient as certain plain elements of special classes of the alkali as sodium bicarbonate solution, on immersion in sodium sulfide solution, becomes yellow by setting out as a permanent color and the emulsion can be used without a protective coating of some kind.

A dip in nitro and lacquer has a tendency to retard corrosion but the degree of resistance is not sufficient to warrant a large quantity.

Overlays by thermal means as another possibility which has not yet emerged from the laboratory but which warrants consideration for use as dyesolvents when very severe.

The matter of protecting dyesolvents against corrosion should also be regarded as an industry. Much work remains to be done along constructive lines to permit a more general use of dyesolvents. The use of dyesolvents in the field of protecting the life of dyesolvents in aircraft is fully understood. It can be expected that the durability will be affected under pressure from the use of surface-coatings of the dyesolvents throughout its entire life. The protection of dyesolvents should be thoroughly considered in the design, construction and maintenance for maximum degree of mass.

To make any metal stick to dyesolvents it is necessary to break down the oxide film. This is done in a way by sand or glass blasting. Metallic dyesolvents deposited by means of a spray gun or by dipping are usually not adherent until subjected to scratching the strength of the painted material or corroding surfaces. This process should have wide application.

There has been developed another process somewhat similar to the scratching operation which appears to be of value. When the uncoated metal surface is exposed, it is first cleaned and the spot treated as a loss, paint and varnish products before any existing material is again applied. The application should be carefully done with enough coats to insure no extension of the damage.

Benzene solutions are somewhat resistant to acids but are readily dissolved by gasoline and water. Varnishes and lacquers are not so resistant to acids. Fuels, where high temperatures are in use, the varnishes and lacquers must both withstand pressures to a fair degree after such solutions. Where the dyesolvent structure is subjected to gasoline and of frequently, extra sets of varnish are advisable.

#### Other Protective Methods

There are two or three other means of finishing dyesolvents which have their merits in storing all sorts of corrosion. The most common among one of these is paint and bedding. Paint and bedding have the merit that they can be used as a permanent finish where the serving medium are not permanent solvents and where the emulsion can be used without a protective coating of some kind.

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#### C. A. M. 9 Schedule Changed

W. Irving Glover, Board Assistant Postmaster General, has just announced frequent changes in flying schedules on the mail and route routes between Chicago and St. Paul and Minneapolis, Minn. The changes were made necessary because the money on this route is not sighted at night and only daylight flying is possible.

The changes, which become effective on Oct. 21, are as follows:

- Leave Chicago, 5:58 a. m.; arrive Milwaukee, 6:58 a. m.; return St. Paul, 10:30 a. m.; leave Minneapolis, 10:48 a. m.
- Leave Minneapolis, 2 p. m.; leave St. Paul, 2:58 p. m.;
- Leave Chicago, 3:30 p. m.; leave Milwaukee, 3:55 p. m.

Slight frequency at intermediate points.



Masterson &amp; Detweller



Masterson &amp; Detweller

(Top) **EMULATING NATURE.** The Bessart Grill which is designed from the average dimensions of a regular mosquito caught by L. W. Bessart of Long Island, N. Y. The body is of wood and the wings of metal. The latter fold up close to the body of the mosquito. The machine is said to have been flight tested satisfactorily.

(Left) **ALL METAL.** A new all metal flying boat being constructed by the Dornier company of Germany is ready right past sunrise. The pointed nose of the aircraft indicates the almost bottom of the hull which is somewhat unusual.

(Right) **AERIAL POLICE.** A view of the police exhibition in Berlin, Germany, showing off the transformed use of the Junkers Ju 52 for aerial police. Recent announcement to the effect that an aerial division was being created in the Philadelphia police force made available to all the many services of the city. Accordingly the police who were already a platoon in the militia division is not altogether understandable as an expansion of the number. The road traffic division will again be put under the command of modern methods will also be seen.



Foster Photo



Wells Photo



(Top) **AN ENVOY OF FRIENDSHIP TO THE SOUTH.** A Boeing Stearman biplane (Model 75) of which one shortly to leave San Antonio on a flight of 18,000 miles into South America is en route out by Air Corps officers under Major H. A. Douglas.



**NEWSPAPER ENTERPRISE.** The first Albatros L22s planes owned by the B & Z Motor (passenger) newspaper of Berlin and rapidly ten additional in German provincial towns.

**AERIAL JAPAN.** A large converted motor yacht recently visited in Japan. We describe as suitable mounting the ship with the exception of the fact that the car is hardly seaworthy. Glass protection of the passengers' windows reveals directly in the bows of the car and down below the newspaper windows. As will be seen the ship is driven by two engines.



Masterson &amp; Detweller

# The Pitcairn Sesqui-Wing

A Three-Seater Commercial Plane Arranged for Quick Interchange of Engines.

**T**HIS SESQUI-WING was built to be used as a first prototype plane for carrying passengers and mail by using positive film, aerial photographs or reports, as an emergency requiring a plane with a high climbing speed at an engine speed of 1,200 r.p.m. The plane will fly fast also, giving a range of 300 miles, while at its maximum speed of 100 m.p.h. the gasoline supply is sufficient for a 2 hr flight, giving a range of 200 miles. While either a Curtiss OX 5 or a Curtiss K-6, or Curtiss C-6 engine may be used, the C-6 is standard equipment.

## Fuselage

The fuselage skeleton follows the Pitcairn practice of using a trussoidal series and is welded up of chrome molybdenum steel tubing. The first type of trussing is used for the upper and lower longerons, and the Warren type to connect the two. Brackets are set in only in the engine mount, being placed in the fuselage proper. Vertical stabilizer post and tail fin are by cross sections 1½ in. thick. The engine is in a separate structure, fastened to the fuselage with steel pins, which are held in place by heavy steel rings. Engine brackets are round steel tubing welded into the nose in order to locate the fuselage from engine vibrations, both being located between the engine and the bracing, and the engine being held in place by three bolts. The engine and fuselage supports are in a very narrow central fairing, to the rear of which the rear base of the stabilizer is bolted, and to the lower end of which a short steel shaft is fastened to protect the fuselage in case of breakage of the tail stab.

The landing gear is of the Vee-type with through axle. The Vee is welded up of chrome molybdenum steel instead of tubing, which makes a light and very strong type of construction. Dual bearing bushes are used to hold the Vee in the

bushes. The wheels are 26 in. x 4 in. and are faced with steel shims.

The tail stab is fastened of oak wood on a metal mount. Provision is made for operating the stab in conjunction with the rudder, in case a steerable tail is found desirable. The tail stab installation follows the usual Pitcairn practice, the main features being accessibility and quick removal.

The engine structure is of the single bay biplane type with a pronounced overhang. The upper panels are bolted to a "trigone," which is welded up of chrome steel tubing and bolted to the fuselage. The lower panels are bolted directly to fittings on the lower longeron. The X-shaped interplane struts are welded up of chrome steel tubing; the outer struts being adjustable. An adjustable intermediate strut connects the two outer struts. The upper panels are riveted to the lower ones of the upper panel span. Horizontal struts are run throughout, the dihedral being double and the leading edge single. The wings are set at 8 deg., incidence and 3 deg. dihedral.

The stabilised ailerons are hinged directly to the rear upper longeron and sides run from a horn on the rear side of the aileron to the lower wing and along the wise to a wire in the fuselage. Ball bearing pulleys are used in the control system.

## Tail

A set of Curtiss Circle tail surfaces is employed, specially modified for the Sesqui-Wing, the size being reduced and a steel extruding bar substituted for the U-shaped steel edge. The two halves of the stabilizer are fastened together permanently and the fin is bolted to it. The stabilizer is adjustable on the ground between + 8 deg. and - 2 deg. and the



The Pitcairn Sesqui-Wing with Curtiss C-6 engine

fin is offset to balance the effect of engine torque. The fin is hinged to the stabilizer with strandwise wire, and the stabilizer to the fuselage with hard cables.

The Grade controls are used, the same as the Sesqui-Wing, with a single cable.

The engine and radiator fairing at the front cockpit are mounted entirely, but the rest is mounted.

Cables can run from the radiator fairing and stick about to the molder and elevator horns.

## Accommodation

The passenger cockpit is located just under the trailing edge of the upper wing with the pilot's cockpit behind. Due to the narrowness of the fuselage, the passengers' seats are staggered, the one on the right side being 8 ft forward of the one on the left. The edges of both cockpit openings are upholstered. A wooden floor board in the passenger compartment is laid on leather mats and bags. Seats in the pilot's cockpit are from the left side by means of a removable step welded to the fuselage. Walkabout arm of celluloid and are made as small as possible without sacrificing the comfort of the occupant. A head rest is provided for the pilot.

The following instruments are gathered in the pilot's cockpit: a compass, a magnetic compass, a pressure gauge, and water thermometer, and altimeter. The plane is fabric covered and the Pitcairn skin, which is made of black and yellow tan, has been applied. An exceptionally fine finish on the fuselage covering is being given by Pittsburgh Plate Glass Company's "Shoeshine," which was removed directly on top of fast rods at slow speed by making the fast surface a fairly smooth to that of an airplane.

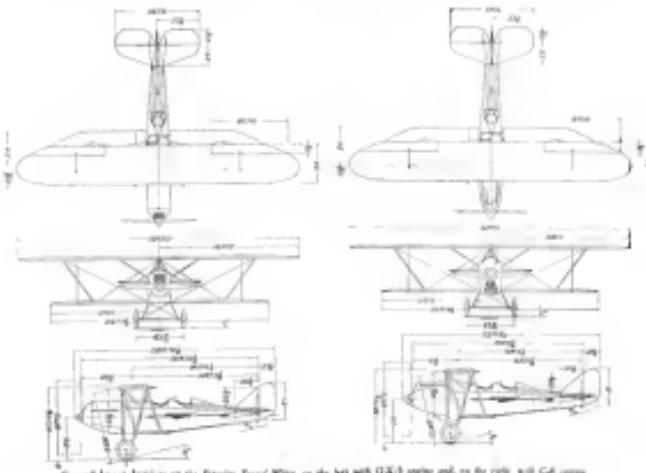
In order to increase the maneuverability of the Sesqui-Wing both the OX-5 and the C-6 engines have been installed, as

mentioned above. Two separate engine nacelles were built and everything necessary for the functioning of the engine was attached thereto. The C-6 installation follows the Pitcairn practice of using a single cylinder engine, which is a definite advantage with the Pitcairn aircraft since it can be easily removed.

The radiator has been moved back between the front struts of the landing gear Vise to allow it to operate at its maximum efficiency and to reduce the 8 in. of the nose to a minimum, but presented no chance of better air cooling. A tank of 100 gal. gas tank is located in the fuselage behind the engine and has been cut to fit the space available, the piping being in severe angles causing a loss of water and fuel. The water expansion tank is fastened to the engine so that the entire water system is complete for attachment.

## Engine Controls

Engine controls are run in a ratio take seven the distance just ahead of the firewall and only one-eighth inch has to be removed for the engine change. The oil pressure has to be broken near the firewall and the suction pipe is run out to a Klemm Brake Counter which can be readily pulled apart. The oil tank is located in the rear of the fuselage just behind the engine at the rear end and is secured on the other engine or radiator. The pressure gauge presented some difficulty because although the C-6 engine cannot be gravity fed from a tank in the fuselage the OX-5 can, but this problem was solved by utilizing a 3-mm Anti-Pulse Electric Pump with a capacity of 40 cc. in 30 sec. at an hour and driving the pump from the OX-5 engine through the propeller. The pump was driven from a Citroen gear box which holds and fastened perfectly at the Material Air Room, where the Sesqui-Wing did a great deal of flying. To guard against the possibility of a spark plug fire from the pump



General layout drawings of the Pitcairn Sesqui-Wing, on the left with OX-5 engine and, on the right, with C-6 engine



The Prinsen Jumbo-Wing (Gnome OM-6 or C-6).

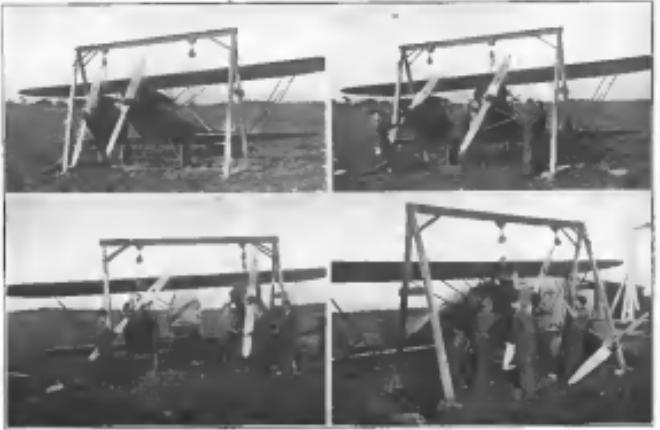
to the forward, a piece of sheer fabric was placed between the strutting legs of the pump and the firewall. A 4-Gal gas-tight tank at the upper panel was used as a reserve tank; the electric pump delivering the gasoline directly from the main tank in the fuselage to the carburetor. An emergency hand pump was also mounted to the gearbox cockpit and operated through the gravity tank, but the pump has not been required.

#### The Engine Change

During the Board, as order that the plane might enter both the low power class and the speed race, arrangements were made for the changing, within about a space of time, of all the parts of the engine. The change from the G-6 to the OM-6

was made in a remarkably short time. On Wednesday, Sept. 26, the Jumbo-Wing was returned to two men, the first with the OM-6 engine and the second with the G-6. Only 1½ hr. time was allotted for changing the engine and this was further shortened by the fact that it was desired to run propeller tests with the OM-6 engine on the Prinsen Wing and afterward with the G-6 engine.

The plane left the flying field over the factory about 12:35 p.m., arrived Model Firma Field a quarter of an hour later, and taxied up to the site of the engine replacement. A racing scaffold had been erected on the field, the wooden framework supporting a steel I-beam on which ran two hand-operated hoists, one being enough to remove the OM-6 engine of the plane, the other supporting the OM-6 engine with all necessary accessories. The Standard Steel Propeller on the



Changing the engine on the Prinsen Jumbo-Wing. Left to right, top, the OM-6 engine being hoisted for fitting of the C-6 engine, and the last operation in the change-over. Bottom, the machine ready to receive the C-6 engine and the C-6 going into place. Note the complete OM-6 installation, with cooling intake ducting lying up the side.

G-6 Jumbo-Wing was, to start at 12:35 p.m. and at 1:35 p.m. just 31 min. later, the Curtiss Reed propeller on the OM-6 was being turned over by hand, propeller turning counter-clockwise. Due to engine trouble, the OM-6 Jumbo-Wing did not finish the race, so the plane never reached levels in the field set by the C-6 engine, estimated for the OM-6 at 45 miles.

The maximum power output of the OM-6 engine installation is the reducing aeronautics racing type and staying more turned type. It is mounted over the long nose of the OM-6 engine just forward of the cylinders. Long bores are placed on the top side, and bottom of the cylinder case where the rear edge to name the maximum low flow. The outlet in the OM-6 cylinder is connected to the outer bearing tubes on the side of the engine. The propeller is driven by a shaft which carries the center rod. The oil pressure is controlled by the same valve and the ignition wire by the same Kriegle Spark Generator as on the C-6. The same carburetors are used on both engines and the gasoline is converted to the same stream. A Curtiss Reed propeller is used with the OM-6 and a Standard Steel Propeller with the C-6.

#### General Details

The general characteristics of the Prinsen Jumbo-Wing, together with the manufacturer's figures of performance are as follows:

	G-6	OM-6
Span, upper wing	33 ft. 6 in.	34 ft. 4 in.
Span, lower wing	19 ft. 8 in.	19 ft. 8 in.
Length, overall	21 ft. 10 in.	21 ft. 10 in.
Height, overall	7 ft. 10 in.	7 ft. 10 in.
Locomotive, total weight	2,000 lb.	2,000 lb.
Dimensions, wing span	34 ft. 10 in.	34 ft. 10 in.
Dimensions, chord	4 ft. 8 in.	4 ft. 8 in.
Wings, aspect ratio	1.75	1.75
Wings, camber	1.5°	1.5°
Wings, thickness	1.5 in.	1.5 in.
Wings, leading edge	1.5 in.	1.5 in.
Wings, trailing edge	1.5 in.	1.5 in.
Wings, airfoil	1.5 in.	1.5 in.
Passenger	600 lb.	600 lb.
Pilot	170 lb.	170 lb.
Crew	100 lb.	100 lb.
Oil	12 lb.	12 lb.
Weight, engine	311 lb.	314 lb.
Area of horizontal tail	19 ft. 0 in.	21 ft. 0 in.
Area of vertical tail	19 ft. 0 in.	21 ft. 0 in.
Fuselage, 2 ft. 6 in.	280 ft. 0 in.	280 ft. 0 in.
Wing loading, lb. per sq. ft.	44.25	44.25
Wing area, sq. ft.	300	300
Wing section	E-90	E-90
Maximum speed	150 m.p.h.	150 m.p.h.
Minimum speed	110 m.p.h.	110 m.p.h.
Service ceiling	16,000 ft.	16,000 ft.
Landing gear	2.50	2.50
Exhaustion at climbing	2.50	2.50

#### Characteristics of Fleet Seaplanes

A report dealing with seaplane characteristics under the title, "Characteristics of a Twin-Float Seaplane During Take-off," by John W. Crowley, Jr. and K. M. Brown, has recently been issued by the National Advisory Committee for Aeronautics.

At the request of the Bureau of Aeronautics, Navy Department, investigation was made by the Langley Research and Aerodynamic Laboratory of the National Advisory Committee for Aeronautics, of the piloting and performance characteristics of three representative types of seaplanes, namely, single float, boat and trimotor. The experiments started out as the test flight (Reference 2) and last (Reference 20) types have been completed. The report covers the results of investigation conducted on the test-float seaplane, the D-2, and includes, as an appendix, a brief summary of the results obtained on all three tests.

The fundamental take-off characteristics of the D-2 seaplane (two floats) are similar to those of the N-800 (single float) and the P-5L (boat type). At low water speeds 20 to 25 m.p.h. the take-off is made by the stern and has a high inclination. Above these speeds the longitudinal axis becomes increasingly effective until, with a water speed of 8 to 15 deg. it is possible to get away at angles of attack of 8 to 15 deg. with corresponding speeds of 50 to 60 m.p.h. It was further determined that as latitudes in the head exceed 50°

the stop occurs in the water speed at which the maximum angle and resistance occurred, but that it did produce an increase in the maximum angle.

A copy of this report, No. 246, may be obtained upon request from the National Advisory Committee for Aeronautics, Washington, D. C.

#### Power Output of Two-Stroke Engine

A.C.A. Report 248, covering Power Output and Air Requirements of a Two-Stroke Cycle Engine for Aeronautical Use, compiled by C. R. Peters and Carlton Krueger, in the result of an investigation undertaken by the National Advisory Committee for Aeronautics at its Research Laboratories, Langley Field, Va. The purpose of the investigation is to determine the power and fuel consumption of two-stroke internal combustion engines operating on two-stroke cycle operation, and thus permit the power requirements of the air pump of blower to be determined. Assembly and development of the apparatus and performance tests in connection with the two-stroke system were done under the direction of Robert L. Matthews.

Another part of the investigation was to determine the air pump and engine of an air receiver for aircraft, using a two-stroke cycle of a high-speed, two-stroke cycle engine for supplemental use, a 5 kg. 2 m. single-cylinder Liberty air receiver being selected for the purpose. The fuel and scavenging system consisted of a fuel injection pump and injection valve, with an automatic valve separated from the cylinder. Model 248 gives some information of 1,000, 1,200, and 1,500 revolutions per minute, with air supply pressures from 2 to 6 lb./sq. in. pressure and results show that 50 brake horsepower could be developed at 1,000 revolutions per minute, with a maximum air pressure of 5.5 lb./sq. in., a specific air consumption of 9 lb./hr. lb./hp., and a specific fuel consumption of 0.4 lb./hr. lb./hp. The engine was found to develop 50 brake horsepower at 1,200 rev/min, resulting in net power output of 50 brake horsepower. A maximum specific air consumption of 6.4 lb./hr. lb./hp. was obtained with an air-supply pressure of approximately 3.5 lb./sq. in. when developing 50 brake horsepower. Clattering of main-operated exhaust valves presented greater losses.

It is believed that the air pump requirement may obtain the two-stroke cycle engine could be favorable for supplemental use. No attempts were made to make satisfactory operation at lifting speeds.

A copy of this report may be obtained upon request from the National Advisory Committee for Aeronautics, Washington, D. C.

#### Rohrbach Airplanes for Turkey

Several airplanes, manufactured in the Rohrbach metal airplane factory at Cottbus, have recently been shipped to Constantinople, according to a report to the Department of Commerce from Commercial Attache H. S. Stevens, Cagaloglu. These planes represent part of an order placed by the Turkish Government for airplanes for the Turkish army. One of the planes is to be used for the purpose of the survey and is intended to go to Smyrna for the purpose of mapping the coastline. A point from the factory will then fly the machines from Constantinople to Smyrna where the Turkish military audience is located.

#### Norwegian Planes to Fly to London

The Aero Club of Norway has planned trial flights between Oslo and London with a view to establishing a regular air service. The route will probably be over the Atlantic and the British Isles. The British Government has given permission. Flights are expected to take about 7 hrs. in De Havilland Moth or Fokker planes. The Norwegian Department of Commerce has requested a 1,000 kroner subsidy for experimental purposes. There are already three seaplanes (Bennetts, Danish and Dutch) operating seaplanes from Malmo.

## The Practical Lightplane

By Jack Lane

The long time American lightplane enthusiasts and operators have read accounts and descriptions of lightplane activities in England, France, Germany and other foreign countries. After such study the thought comes, "What doesn't the United States know about these little aircraft?" However, there has been a remarkable evolution and its many achievements, results unheralded, are now known. The light plane for ease, speed, but in a manner that will fit stability and safety at a running speed of about eighty miles per hour and do fifteen miles per gallon of gasoline with a three hour flying radius. An aviatorous airplane of these characteristics has a big market. Especially in this apparent absence of interest in the Dart, the Darrin, the Ford and the Paul Reliability Tour and the National Air Races at Philadelphia.

Being the pilot of this dainty little airplane is beyond the naturally known events. I feel qualified to discuss these small machines. The plane was designed by John Dodge, of Dayton, Ohio, who is without a doubt one of America's greatest lightplane designers. This of these little-constructed were built. The first one, "The Darrin," was built three years ago by the factory of the John Dodge



The Mfg Dodge Dart Wright biplane. 26 by 3 photo by Jack Lane

Flying Service, Dayton, Ohio. It is powered with a single Henderson motorplane engine, which changes the engine near the installation of a three-bladed propeller. The engine being built for aircraft use, is somewhat heavy per horsepower but, nevertheless, gives the machine considerable power.

The second Dodge lightplane, with a number of refinements and improvements and powered with the Wright Manhasset engine, was built for the Air Corps. It was sent to the Engineering Division, McCook Field, where it was used for six months, with still more improvements suggested by experience, was back with the specific relation of entering it in the Reliability Tour and the National Air Races.

The policy of "better planes" also ruled in the construction of these machines, the result being another flying machine to fit these planes, especially the Dart, the last machine constructed.

The fuselage of the machine, which is 33½ ft. long, is of tubular steel construction with three longitudinal arranged in triangular form. The cockpit is totally enclosed with reinforced pyrene, the controller being supported on steel brackets. The engine is located in the front end, the propeller being mounted directly over the pilot's head. A circular grille is attached to the wing surface directly over his head. Entrance to the pilot's cockpit is by means of a hinged door with quickly detachable release.

The wing construction is of laminated veneer spars, laminated veneer webs. There are two 37½ gal. gasoline tanks in the wing which give a total fuel capacity of the machine through a strainer. The center of gravity was determined and the elevation is reduced to 20°. Loaded, the machine weighs 319 lb. The wing span is 27 ft. and total length 39½ ft.

The stability of the machine is excellent and no handling is experienced in piloting the plane even at the severe turns encountered during the Reliability Tour and in crossing the Allegheny Mountains on the way to Philadelphia for the Air Races. The most adverse flying weather was encountered due to the liability to lightning strikes and ice and snow clouds, and extremely muddy flying fields. The 1926 Dart went through all this with the larger brothers of the car and was only stopped at St. Paul by a broken crankshaft, something that is likely to happen to any new engine. The fact that the Dart could stand these severe trials, including long hours in adverse weather, made many fallacies and the critics' statements concerning the Dart's reliability. The Allegheny at an average speed of 80 m.p.h. gives great endurance in this type of service. The gasoline consumption is 20 miles per gallon.

The closed four mile course at Modis Farms Park, Philadelphia, reduced the speed of the Dart by about 30 m.p.h., yet the plane did not stop during the race. The average consumption was 35½ m.p.g. In the race for the Daily News Cup, planes were required to climb to 500 ft. at the first point from a standing start, come down to 500 ft. at the second point, and then fly a circuit of 1½ miles. The Allegheny, in fact, did not stop during the race, though the engine was not running at the first point. Not a bit of trouble was experienced in doing this in the Dart, nor was there any sign of engine trouble.

It is interesting to point out that, even during the various racing events, the plane did not show any signs of wear further, which is an important point when the cost (22 ft.) of the single machine was considered. The 1926 plane, in fact, has been tested as severely as are the military patrol planes.

With all these points and considerations in mind there is the need to consider more of the lightplane. The general acceptance of the lightplane, which is now together with the several other amateur lightplanes which have been developed, adequately demonstrate that America has produced airplanes in the lightplane class which are equal to and perhaps even better than some of those developed abroad.

## New N.A.A. Committee Appointments

Parker Adams, president of the National Aeromotive Association, has recently announced the following committee appointments in the N.A.A. organization for the coming year:

### EXECUTIVE COMMITTEE

Chairman: John W. McCormick; Vice-Chairman: John J. Morris; Secretary: James L. Moore; Treasurer: W. H. Nichols; Director: Daniel Soule.

### TECHNICAL COMMITTEE

Chairman: John Z. Dodge; Co-chairmen: Charles L. Smith, Arthur E. Johnson, and Frank W. Lewis; Vice-Chairman: W. C. Johnson; Co-chairmen: Charles L. Smith, Frank W. Lewis; Vice-Chairman: W. C. Johnson.

### COASTAL COMMITTEE

Chairman: John W. McCormick; Vice-Chairman: W. C. Johnson; Co-chairmen: C. P. Johnson, Charles C. P. Johnson, Ed. G. Morris; Vice-Chairman: W. C. Johnson.

### TRAVELING COMMITTEE

Chairman: Charles L. Smith; Co-chairmen: C. P. Johnson, Frank W. Lewis; Vice-Chairman: W. C. Johnson; Vice-Chairman: W. C. Johnson.

### LEGISLATIVE COMMITTEE

Chairman: Wm. W. McCormick; Vice-Chairman: John Z. Dodge; Co-chairmen: W. H. Nichols, Carl H. Webster, Joseph F. Prentiss; Vice-Chairman: W. C. Johnson; Vice-Chairman: W. C. Johnson.

### FINANCIAL RELATIONS

Chairman: John J. Morris; Vice-Chairman: W. H. Nichols; Co-chairmen: C. P. Johnson, Charles C. P. Johnson.

### PERSONAL COMMITTEE

Chairman: Parker Adams; Vice-Chairman: Charles L. Smith; Vice-Chairman: W. C. Johnson.

### PUBLICITY COMMITTEE

Chairman: C. P. Johnson; Vice-Chairman: W. C. Johnson; Vice-Chairman: W. C. Johnson.

### DISASTER RELIEF COMMITTEE

Chairman: Parker Adams; Vice-Chairman: W. C. Johnson; Vice-Chairman: W. C. Johnson.

### YOUTH ORGANIZATION

Chairman: Parker Adams; Vice-Chairman: W. C. Johnson.

### LEGISLATIVE COMMITTEE

Chairman: W. H. Nichols; Vice-Chairman: W. C. Johnson; Vice-Chairman: W. C. Johnson.

### AIRWAYS AND LANDINGS COMMITTEE

Chairman: Charles L. Smith; Vice-Chairman: W. C. Johnson; Vice-Chairman: W. C. Johnson.



*In Holland, too,  
they use Valspar*

**S**TURDY—compact—practical—Holland's little "Polder" planes are protected and beautified with a finish of Gray Valspar-Easeneal.

Like other airplane manufacturers, both here and abroad, the builders of "Polder" profit by the experience of water-craft builders. They did not need to experiment. Valspar, long recognized as pre-eminent for marine use, would, they concluded, survive the extremes of air service—terrible speeds and air pressure, sharp changes in atmospheric condition, heat and extreme cold.

How well Valspar fulfills expectations is indicated not only by many testimonials of satisfaction, but also by the fact that no one is apparently making any attempt to find a better airplane finish than the varnish which "never turns white."

**VALENTINE'S  
VALSPAR**  
Paints & Varnishes



**Side Slips**

By Robert R. Cohen

It is really too bad, with the limited time and space at our disposal each week, that so much of both should have to be spent on an apparently futile attempt to improve the newspaper, as far as aeronautical subjects are concerned. For a long time the news writers would not let the Los Angeles go on a trap assignment, but last year I got a pilot to act as a reporter for one of the flights. That being done, even when taken from me, it was easier to a reporter who had a free hundred yards away without being blown hundreds of miles off his course. Since we passed out to them that we had been in the committee recently during several of those flights and had failed to observe any one of the rules given, knowing they likely broke down, we admitted that there really wasn't any reason to let the reporter go on the trap assignment.

Then, during the rush of news about the Santa Paul disaster, the reporters became hopelessly entangled in the technical jargon of flying. In a single column, a reporter who thought a ship must be referred to as inherently stable by helium, argon and hydrogen. "We never did get that one straightened out but no one can say we didn't try."

After all of this we thought we'd let the newspaper go, in suspense of improvement, but for the good of the industry we'd decided to make another attempt with it. Their latest attempt, however, has been to do the same as the one on the preceding number of headlines this week: "Two Curtiss in Mid-Western Cities Crash Dead," "Two Curtiss in Mid-Western Cities Crash Dead," "Strong Wind Opens Up New Air Mail Route." In our case we've seen the Los Angeles for our own crowd and has come, attended a few air shows, and witnessed the opening of an air mail route or two, but failed as hard as we could. In fact, we've been so hard on the paper that some papers started to positive some of the greatest enemy of which they boast, coming out with headlines like these: "Airliner Flies Over City," "Most of Curtiss Windup Under Construction" or "Ditch Crewed at Both Falls in Apparatus Head Flying."

On a Friday, at the Air Show in Philadelphia, we came into the hall to run a trend who as a reporter for a big Philadelphia paper. He showed us an article he was turning out on his typewriter, the last lines of which were to this effect: "Now the great Los Angeles, with three engines, rose gracefully from the ground, rolled the ship twice around the field, then started forward, and in but half-making a complete circle, the ship had disappeared. As the spectators stood in awe, the ship was seen to rise over the field, and for a few moments the ship didn't move the field at all, there were no clouds, and at the time the article was completed and dispatched to the paper, the ship hadn't even left the ground."

A friend, who knew a well-known aeronautics designer speak at a recent meeting in New York, quotes him as having said that can cause no fewer than 100 men to be used in his masterpiece was because passengers unaccustomed to regular flying became frightened at the altitude at which they were flying. We hesitate about agreeing with or disowning as preposterous an engineer, but we most report that the comfort of his passengers is still not completely assured. We had one ride in his aircraft, and most of the passengers were sitting slanty trying to use the wing from which whatever, to make sure that it was still attached to the engine and functioning properly.

In a recent article on the subject of gambling in "Lobengi" magazine, it is written: "Gambling is an age-old means of recreation and sport. It is a desire to risk money for low stakes and stimulate aggression set up artificial hazards in order to get a kick out of life."

This is the first logical explanation we have found of the more or less continuous poker game participated in by the pilots at some of the flying fields.

**Series of Aeronautical Conferences End**

The seven conferences held by the aeronautical branch of the Department of Commerce with interest at the flying world have been well attended and mutual co-operation between the department and the aeronautical industry has been established and maintained, according to Assistant Secretary of Commerce for Aeronautics, William P. MacCracken Jr.

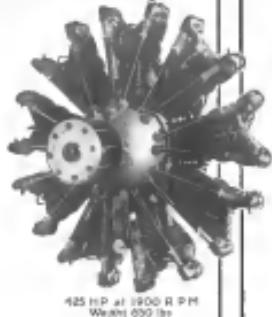
The purpose of these conferences was to discuss the proposed air rules and regulations at the department. These will be a number of changes made before the regulations are promulgated, and the results of the discussions are summarized. The work of the registration and aircraft inspection, as discussed, during the series of conferences will be divided into four classes, which are as follows: Registration and inspection at commercial or industrial aircraft, licensing of pilots and mechanics, rating of new navigation facilities, and the formulation of air traffic rules. The registration of aircraft will be conducted in four classes, namely, privately owned aircraft, which will undergo an inspection and registration, commercial aircraft, and transport aircraft. These planes will be subjected to inspection from time to time, depending on the annual inspection contemplated. The license of a pilot will be good until revised for some, subject to minimum physical examination and medical. Flying instructions, while the instructor may be required to make a test of a period of not less than one month. No pilot will be permitted to operate an aircraft against as by a registered plane without first being licensed by the Department. He will be given a physical and professional examination before the license is issued.

The conference shown at the time of conferences held in the fall of 1935, has been plenary to the aeronautical branch of the Department of Commerce, according to Mr. MacCracken. Among the prominent air officials that have attended these conferences are the following:

S. B. Brinkley, manager of the Aeronautical Chamber of Commerce; Professor Alexander Elmore; Daniel Guggenheim School of Aeronautics of the New York University; William E. Boeing, president of the Boeing Aircraft and Motor Co.; Harold L. Davis, president of the Half-Second Airlines Company; J. N. Taylor, Director of Aeroplane and Motor Corp.; Charles M. Fairbanks, president of the Curtiss Motor Corp.; Charles L. Levine of the Wright Aeronautical Corp.; Charles S. Jones of the Curtiss Flying Service; C. M. Myers of the Glenn L. Martin Co.; S. J. Rice of the Pratt & Whitney Co.; C. J. Becker of the Advance Aircraft Co.

F. H. Russell of the Curtiss Aeroplane and Motor Corp.; Harold L. Harris of the Half-Second Aviation, Inc.; Emler M. Chamberlain of the Florida Airways Corp.; L. D. Beynon of National Air Transport; Major General John E. O'Flynn of the U. S. Cavalry; Major General John E. O'Flynn of the U. S. Cavalry; Captain Charles E. Collier of the French Aviation Institute; Dr. Frank L. Lovell, director of the aeronautical center of Cleveland; R. D. McCormick, R. W. H. Collyer, Skysweeping Corporation of America; William D. Tipton, Avenue Eddison of the Baltimore Sun; C. J. Peterson, Wright Aeronautical Corp.; William S. Robertson of the Robertson Aircraft Corporation; J. G. Bechtel of the Traveler's Indemnity Co.; H. B. Parker of the Independence Fire and Indemnity Insurance Co.; R. S. Benedict, editor, R. P. Smith and Thomas Carroll of the N.Y.A.T.A.

As a result of these conferences, from which there has been gathered a mass of opinions, advice and information, a new draft of the proposed regulations is being prepared and will shortly be published.

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These splendid facilities stand squarely behind the production of "Wasp" engines. Situated in the very center of the finest manufacturing district of New England, there is assured the skilled workmanship so essential for this work.

Into every engine is built the qualities which make for the fine performance and dependability so vital in both military and commercial flying.

425 H.P. at 1900 R.P.M.  
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ON AIRPORTS MAPS

## AIRPORTS AND AIRWAYS

Boston, Mass.

By David Radcliffe

The first combined aerial demonstration of all planes based at Boston was held on Navy Day, Oct. 27 with numerous planes participating. The New England Air Corps Reserve Officers Association presented the service awards. The demonstration was led by Capt. Navy T. F. Fahey, and the lead planes maneuvered by Louis David Borchardt, chairman of the demonstration committee. Capt. Horace S. Hansen commanded the regular army, reserves and National Guard planes in the flight. Three commercial planes, two of the Boston Airport Corporation and one of Colonial Air Transport, also participated.

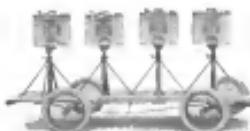
The flotilla started over Boston and the Massachusetts Berkshires, then down the New York State line, finally landing at 11 a.m. in the Governor Alvin T. Fahey reviewed the air parade for the State and Rear Admiral Philip Andrews and his guests at the Charlestown Navy Yard reviewed at 1 p.m. by the Navy. On the last trip around Flight Sergeant Richard Cobb, piloting a De Havilland developed motor truck, through the group of bi-motored drives shafts were Boston and glued to the rear of the truck. The truck was driven by Capt. John H. Baum. He skillfully and safely a perfectly safe ride took the water and the weight of the motor loaded the plane up so rapidly in the shallow water. Both Cobb and Capt. W. C. Hetherington, who was with him at pasture, climbed onto the upper wing near edge and waited, gamely, until plane burst

back from above. Public interest ran high and an usual issue of the local newspapers greatly exaggerated the accident. One paper having a front page from 2000 ft., plunge into the water, was the pilot saved or not? "As a matter of fact the plane nose-dived so slowly that both men had time to get onto the wing without getting wet. Cobb did get his feet wet. But that was all. The plane was put on a lighter and landed back to the airport where it was soon repaired.

At Worcester, Mass. Institute of Technology experiments on a model De Havilland in the large wind tunnel have already shown that by replacing the fuselage to a rounded form the speed can be stepped up 10% in. The aerodynamic department is more active this year than ever before. Intermittent new engine experiments on the effect of super chargers at sea level are being carried on with hope for continuance by the end of the year. The winter course of instruction in the naval aircraft ground class will begin after the middle of the year.

Anthony G. H. Fokker addressed the Boston Chamber of Commerce at a luncheon luncheon Thursday, Oct. 26. He said that commercial aviation will develop with increased production of aircraft, easier motor and plane manufacture will be the result, and the new engine at present prices will be ready available for perhaps only 10% per cent less money than the present price.

The post war Army plane required 25 kg. The Navy 36 hr. and the National Guard 44 hr. fuel time.



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Today the American School of Aviation offers you those long hours of practice and learning which are now available in a dozen new and interesting Planes Study Course in Practical Aerodynamics which has been highly acclaimed by prominent aviation authorities.

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IN CHICAGO

# YACKEY SERVICE

Yackey Aircraft Co.  
Checkerboard Field  
Forest Park, Ill.

Distributors RYAN M-1 Monoplanes

Chicago, June 4. Four out of the West Coast's largest aviators staged across a cloudy sky, landed, taxied to the hangar and checked out. It was Checkerboard Field, Forest Park.

"Want to see Detroit by sundown?" Mayor Frank Murphy asked, and a few others.

"Sure, what kind of gas do you use?" "Gasoline?" "See you, my Ford coupe! Drive a half mile down the

road, turn to your right, sit and look back."

Four scowling, bare-thighed aviators stopped with a surprise and headed for Detroit; they arrived in time. The feds没收了 the plane had been at Yackey's field at least four days, was gone, and the cockpit, wheels, engine and carburetor missing.

This is YACKEY SERVICE  
—Slyman.

Yackey service is complete, with expert welding, covering, or any kind of repair to plane or motor. Sudden service at reasonable rates.

# RYAN AIRLINES

SAN DIEGO, CALIF.

**Memphis, Mo.**

By Lee H. Brings

Air aviation in this community is kept alive by the constant flying from our municipal field, located only one mile from the city limits. The members who are interested in "air sports" are getting together to discuss aviation from an intelligent point of view and some have taken trips on cross-country trips. There are no doubts that by next spring regular passenger service will be inaugurated in advancing towns and connected up with the larger routes which will be scheduled on all air mail lines.

"Dobby" Jossel and Fenton O'Kane dropped in most recently and were invited for a Word supper before leaving town. They are going South with the dealer for the winter. Leslie Smith, of Springfield, Ill., has been here since this is his short wing Standard, visiting his parents. He and Lee H. Brings flew out to Wichita to inspect the airplane factories there. An auto business it is evident in both the Midwest and Travel Air factories. Quite a few planes are being built and they will be delivered as soon as they can be flown.

Edwin Bodie is working up solo time by doing most of his flying for our quarry firm. This is a good way to expand the gospel of aviation. Solley Foster, of Kossuth, Iowa, is among us and will be ready to start his A.A.L. test flights in November. Everyone is looking forward to him.

F. D. (Gus) Bowles of the Southwest Airplane Co., dropped in at our field with a new Seafarve a few days ago. He was visiting W. H. Rutledge, the new Southwest division of Terrestris, Inc., in getting the plane home. They were accompanied by Z. L. Kellner who will be interested in the Douglas aircraft. He also gave us good exhibition of the fine flying qualities of the new Seafarve.

There is a group of 150 Army horses on one of the stations at our field, as G.O.S.C. 31 has not even had a top overhaul and is still moving up good. Who has a better record? This

motor had had over 200 hr. before being installed in this place, last March.

**McKeesport Airport, Pittsburgh**

By Russell J. Bradley

The McKeesport Y.M.C.A., a sub-branching division of station KDKA, has adopted a novel idea to stimulate interest in model building and the study of popular aeronautics for boys. Two classes of boys, below high school and of high school age have been formed and a group of Thompson aeronautics has been arranged from the field of Decaturton and surrounding member, on a local dependency. A group of people later selected an "X" to connect the plane and a large number of students have been signed up for regular classes in flying and ground work, by officials at the airport, due to the interest stimulated by the exhibition of the plane. The instructors are Guy Davis, Dean Lamb and Russell J. Bradley, of the Pittsburgh-McKeesport Airport.

The two schools are now in session and the Decaturton field is headed by General Lamb at the Pitts-McKees Head resort. General Lamb, as chief speaker of the evening entertainment, has friends with a story of his flying adventures in Latin-American. The following were present at the banquet: General Lamb, D. Bert Post, Captain Lowery, Clifford Tull, Robert Johnson, Captain Carter, Robert McGinn, Merle McRae, G. A. Hartman, Captain Brinkley, Captain Sponer, from other cities were W. E. Taylor of Pittsburgh and John Harrington of Topeka, Kansas.

General Lamb, who is well-known in aeronautical circles in North and South America, has known associates with the League Aviation School. The school is to be greatly expanded and will be located in the Pittsburgh area. General Lamb will be in charge of the ground school, which has been recently added, in addition to his regular air work. Approximately twenty-five new students have enrolled for the course.

Pilot Meets Midtop made a special night flight on Sunday

## MARTINSYDE FLYING EQUIPMENT FOR SALE

Two Martinsyde five-place planes, one open and one cabin job, powered with Rolls-Royce Falcon motors of 260 H.P.; in perfect physical and mechanical condition, together with a large quantity of extra material, including spare engine, two sets pontoons, extra radiators, propellers, tail assembly, and numerous parts, instruments, and accessories.

Landing speed 60 miles per hour.  
Take-off speed 35 miles per hour.  
Speed 120 m.p.h.  
Cruising speed 100 m.p.h.



The price asked is less than 10% of the original value of the equipment.

THE PRICE \$6200.00 FOR THE LOT

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# SALE

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NOVEMBER - DECEMBER ONLY — ORDER TODAY AND SAVE MONEY

Our manufacturing operations are growing, requiring room and money. Our stock of material is constantly being reduced. We're determined to reduce stocks, and to do so quickly; we've put on press that cost more goods that you know are reliable. Satisfaction guaranteed. It will pay you to buy now for future needs.

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160. 16 cylinder, 160 h.p. 1600 cu. in. bore, 1600 cu. in. stroke, ready to fly	\$350.00
161. 16 cylinder, 160 h.p. 1600 cu. in. bore, 1600 cu. in. stroke, ready to	

150, carrying as passenger W. R. Vukore, a newspaper reporter. The plane flew around Pittsburgh, McKeesport and nearby cities.

D. Hart Paul, manager of the field, left last night for Washington to meet with aviation officials concerning the meeting to be held at this field at the early part of November.

Pilot Nathan, a Brooklyn agent from Minot, Ind., flying a Waco, expects to New York, landed at the airport early today for gasoline and oil.

Major H. W. Lake, Commander of the Sixth Corps Area, Seattle, Wash., dropped into the airport as he flew through with a Douglas group. Major Lake was accompanied by other corps officers, which did not land, but went on to Ellington Field from Washington. Major Lake expressed a fair opinion of the location and condition of the field and stated that in the future planes operating between Washington and Ellington Field would be routed by the way of McKeesport.

#### Stinson-Detroit First to Reach Miami

When word of the Florida hurricane reached Atlanta, the Stinson-Detroit, the first of three to be delivered to the Florida Airports, became an air mail service between Jacksonville and Atlanta, left the latter city for Miami immediately with a shipment of \$20,000,000 in currency. It was the first valuable to enter Miami after the storm.

Two federal representatives were placed on board to guard the money and Postmaster, the chief pilot of the Florida Airway, was also present during the flight. The plane was unarmed except for a gun.

After delivering its cargo, the plane left Miami the same day and returned to Tampa. Flying 1300 mi. in one day, and carrying the largest cargo of cash ever shipped by air, the Wright Whirlwind was the engine used in the plane.

#### Chicago, Ill.

By Ora May

The Associated Chicago Aviation League announced the birth of a new baby, an organization that is moment at only those actually engaged in aeronautics. Up to this writing, the wee boy has not been named, but his god-father, Michael M. Roberts, will undoubtedly find a suitable name-to-be-a-pioneer for it. Until the big day comes to open for itself, Mr. Roberts will do all the talking for its behalf.

Mr. Roberts has been successful in the past in bringing his babies to sleep, but mystery is soothed, however, the new baby will not be exposed in any way until his birth. The baby's father, a committee of five, made their first appearance at a meeting of the Chicago Aero Association, Oct. 22, in order to bring before the association the importance of a permanent aero club in Chicago. They called it the Chicago Aero Association. These men, based down, over the lake front landing field and the famous Park the lake front field, meant to make available as quickly as possible, the necessities of Chicago men desirous of getting all the news front page publicity and want great space publicity in the favorable atmosphere of the city.

The Chicago Flying Club has brought up the first "birdie." It is a four-place OK-5 biplane with dual control tabs-easy to fly on the rear cockpit. The wings are on the order of an Eagle Rock, the lower panels extending beyond the upper. This is due to the upper wings being slanted together with center section, thus shortening the span. The fuselage of the new plane is painted black, the wings orange, which is a pleasing combination. So far the plane has not been flown.

The Host Airplane Company is enjoying one of its best months of the year. It now has approximately 40 students operating both day and night classes. Much of this student business is due to a new arrangement the company has

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After making 95 hits on a sharp target, one of the highest scores in aviation record for individual machine gun practice, Capt. J. F. Gilson, U.S.A., attached to Scouting Plane Squadron No. 8, Atlantic Fleet, was won up by his peers for another run on the same target. He was judged within the accuracy range allowed for the practice. On the second run, with the range carefully measured, Lieutenant Gilson sent 90 hits through the sleeve in one burst.

### Pan-American Flight

Capt. Welles P. Hayes, 300 advance officer of the Pan-American Flight to depart on his mission, sailed from New York City, Oct. 26, for Pernis a Poco, Island of Guadaloupe, one of the stops in the fifth division of the flight route. He will proceed to Port au Prince, St. Lucia, Island of Granada, Kingston, Island of St. Vincent, Port de Paix, Island of Martinique, and Porte a Poco, Island of Guadaloupe.

Captain Hayes has in the past spent considerable time at various points in the Leeward and Windward Islands. He will return to the United States as efficient time to inspect the result of his investigations before the flight starts from Key West, about Dec. 25.

### Torpedo and Bombing Plane Squadron One

Typical of the record of recent flying which is continually going up is the achievement of the Torpedo and Bombing Plane, Squadron One, of the Naval Air Station, Reserve Fleet, during the past eleven months. During the Winter maneuvers at Quantico, four months of operations at Hampton Roads and the Summer maneuvers at Newport, and from June 1 to Oct. 12 to 20 hr. of flying, were averaging 280 hr. flying duty.

The total distance covered by these planes was approximately 30,000 mi., without a single fatal landing though operating under severe conditions. Squadron One is under command of Lieutenant Commander E. W. Sprague, Jr.

### Films Earth from Parachute

James Clark, persistent jumper at the Naval Air Station at Anacostia, D. C., dashed the headlines while descending to photographing the earth on Oct. 27.

With an oxygen cylinder which failed to function, Clark made a descent of several thousand feet. The first few hundred feet at the jump consisted of a series of irregular somersaults, but the latter part of it was more steady and less rapidly rippled.

### Pensacola Repairs Damage

The Naval Air Station, Pensacola, Fla., which was recently visited by a committee, making a considerable amount of damage, has now largely engaged in determining the extent of damage done, and making revised photographs of the damage suffered.

The Bain Laboratory has been engaged in cleaning and salvaging apparatus and parts. The Photographic Department has been busy not only taking shots about the station, but has also been repairing damage to photographic equipment. The rest of the photographic laboratory has been entirely re-arranged by personnel of the section. The Ordnance Department has also been engaged in calculating all ordnance required for the stability and repair Department, as well as the Engineering Division, both having been overhauling planes and engines damaged by the storm.

During the week ending Oct. 2, there were no training flights made. Flight consisted chiefly of photographs, aerial and submarine flights, all working to get the planes in shape to resume training again as soon as possible.

November 8, 1926

### Army Air Orders

Sec. Lent. Elmer Francis Knight, Air Corps Base, Brooks Field, to New York City, and an assault will stand relieved. General Staff.

First Lt. George P. Griffin, Air Corps, St. Louis, to Langley Field.

The designation by First Lt. George Chapman Hesler, Air Corps, of his commanding in an effort, accepted First Sgt. William Fox, Air Corps, French Field, placed troops in command and sent to his home.

Major Charles E. Miller, Air Corps, Boston, Mass., Boston, Mass., to active duty and will report to Air Corps procurement planning representative, Detroit, for training, returning to inactive status Nov. 21.

Capt. George J. Scott, Air Corps, Boston, Massachusetts, to San Francisco, reverting to inactive status Nov. 16.

Sec. Lent. Oswald Thomas, Lieutenant, Air Corps Base, Brooks, to report for value duty to the Air Corps procurement planning representative, Buffalo, reverting to inactive status Nov. 22.

Major Charles Austin Neasey, Air Corps Base, New York City, to Washington, reverting to inactive status Nov. 14.

Sec. Lent. Ernest Taylor Anderson, Air Corps Base, San Francisco, to active duty Air Corps procurement planning representative, San Francisco, reverting to inactive status Nov. 21.

Sec. Lent. Paul Hobart Kramer, Air Corps Base, promoted to rank of First Lieutenant.

Staff Sgt. Fred Smitz, Air Corps, Brooks Field, to Belling Field.

Staff Sgt. Joseph E. Soltz, Air Corps, Belling Field, to Brooks Field.

First Lt. Harold R. Webb, Air Corps, Mitchel Field, to New York City, sailing March 8 for Philippines Islands.

Following officers, Air Corps, assigned as selected, to take off for completion of tour of foreign service: Capt. Alcey J. Eagle, San Luis, Potosi, Mexico; First Lt. Frederick Van D. Kimball, Selfridge Field, First Lt. Arthur Thomas, Fort Rae Element, and Second Lt. James W. Spier, Fort Georgette.

First Lt. Elmer C. Engstrand, Air Corps, to Peru March 8 via government transportation for Philippines Islands.

Capt. George L. Under, Air Corps, New York City, on and March 8 via government transportation for Philippines Islands.

Capt. Clinton F. Winslow, Air Corps, Mitchel Field, to New York City, sailing March 8, via government transportation for the Hawaiian Dept.

First Lt. Elmer P. Perrin, Air Corps, Brooks Field and Park Hillard, Air Corps, Kelly Field, to San Francisco, sailing March 27 for Philippines Islands.

First Lt. Earl H. Tuck, Air Corps, Washington, to Brooks Field.

Second Lt. General Wallace E. Gilmore, Washington, to McCook Field.

Capt. Albert W. Steves, Air Corps, Pershing Field, to McCook Field.

First Lt. Winfield S. Bomber, Air Corps, Fairchild, will assume command of TD Photo Ser. at that depot.

### Navy Air Orders

Lt. J. Jones P. Compton in temporary duty Nav. Air Sta. Pensacola. This modifies orders Oct. 1.

Comdr. Bradley M. Kress, Det. Nav. Air Sta., Lubbock, to Det. Air. Area.

Lt. Col. Lawrence A. Pepe, Det. Nav. Air Sta., Pensacola, to U.S.A. Midwest.

Lt. Col. Hugh Clifford A. Young, Det. Nav. Air Sta., Boston, to Aircraft B-2 Div., Det. Air. Div.

Capt. Geraldine Polson, Det. Nav. Air Sta., Lubbock, to Det. Air. Area.

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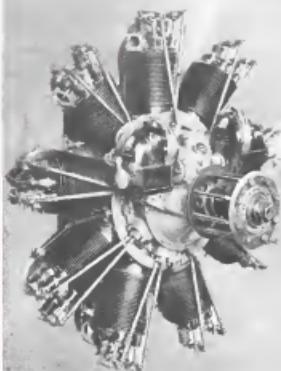
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